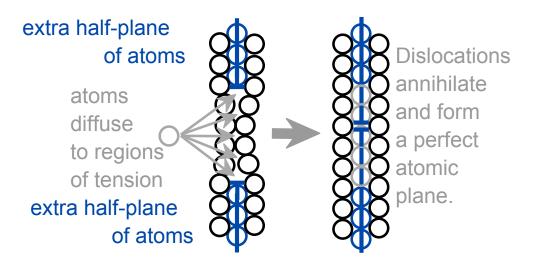
### Recovery

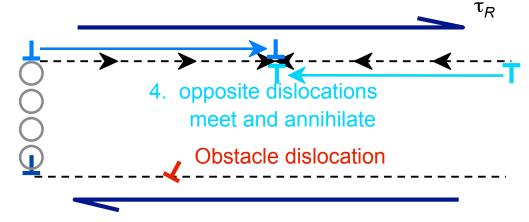
**Recovery**: Rearrangement of dislocations, annihilation reduces dislocation density.

Scenario 1
 Results from diffusion



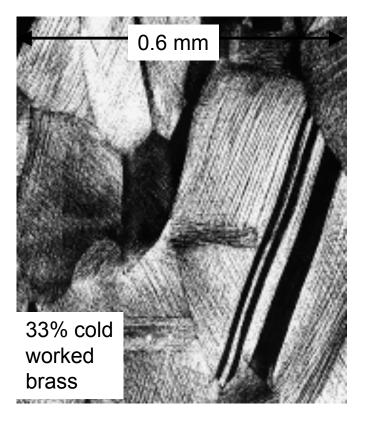
#### Scenario 2

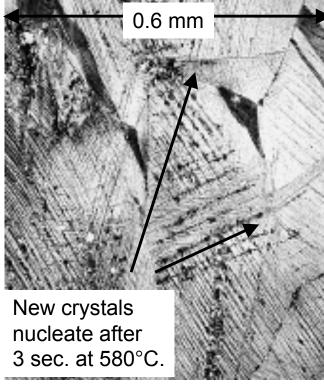
3. "Climbed" disl. can now move on new slip plane
2. grey atoms leave by vacancy diffusion allowing disl. to "climb"
1. dislocation blocked; can't move to the right



## Recrystallization

- New grains are formed that:
  - -- have a small dislocation density
  - -- are small
  - -- consume cold-worked grains.

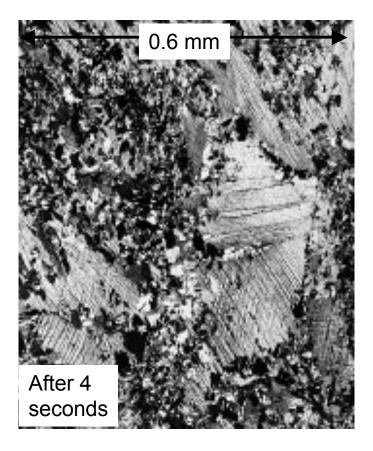


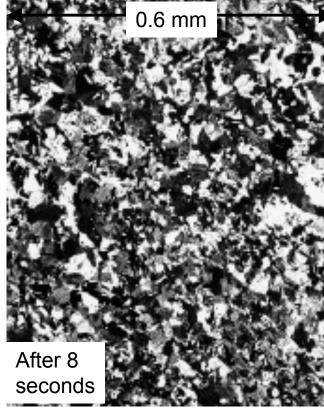


Adapted from Fig. 7.21 (a),(b), Callister 7e. (Fig. 7.21 (a),(b) are courtesy of J.E. Burke, General Electric Company.)

## Further recrystallization

• All cold-worked grains are consumed.

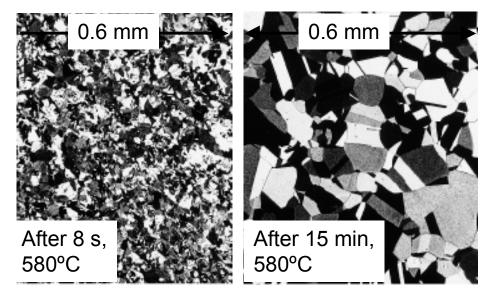




Adapted from Fig. 7.21 (c),(d), Callister 7e. (Fig. 7.21 (c),(d) are courtesy of J.E. Burke, General Electric Company.)

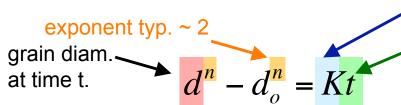
### Grain growth

- At longer times, larger grains consume smaller ones.
- Why? Grain boundary area (and therefore energy) is reduced.



Adapted from Fig. 7.21 (d),(e), Callister 7e. (Fig. 7.21 (d),(e) are courtesy of J.E. Burke, General Electric Company.)

Empirical Relation:

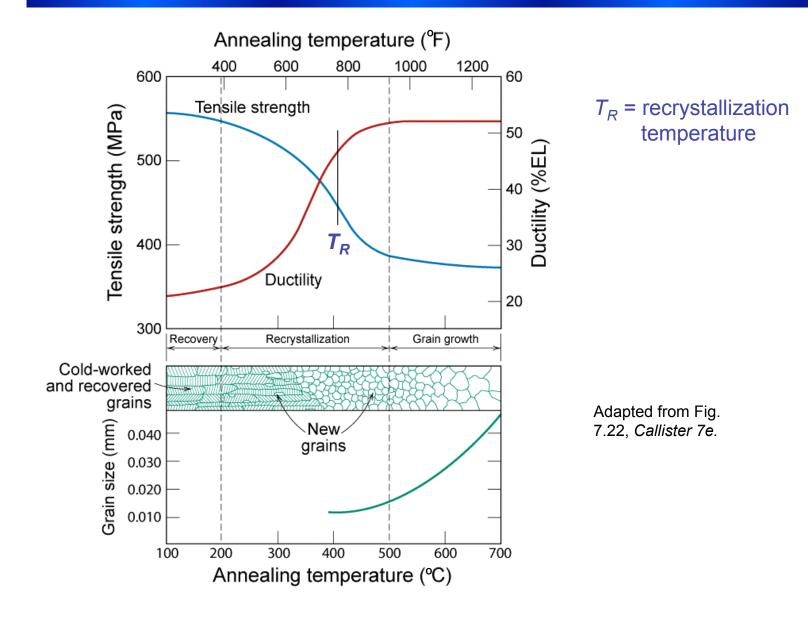


coefficient dependent on material and *T*.

elapsed time

Ostwald Ripening

## Influence of temperature on strength and ductility



## Influence of temperature on strength and ductility

 T<sub>R</sub> = recrystallization temperature = point of highest rate of property change

$$T_m => T_R \approx 0.3-0.6 T_m (K)$$

- Due to diffusion → annealing time → T<sub>R</sub> = f(t) shorter annealing time => higher T<sub>R</sub>
- Higher  $%CW => lower T_R strain hardening$
- Easier to move in pure metals => lower  $T_R$

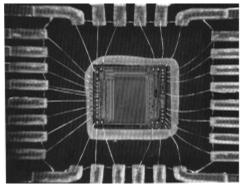
### Mechanical failure

- How do flaws in a material initiate failure?
- How is fracture resistance quantified; how do different material classes compare?
- How do we estimate the stress to fracture?
- How do loading rate, loading history, and temperature affect the failure stress?



# Ship-cyclic loading from waves.

Adapted from chapter-opening photograph, Chapter 8, *Callister 7e.* (by Neil Boenzi, *The New York Times.*)



Computer chip-cyclic thermal loading.

Adapted from Fig. 22.30(b), *Callister 7e.* (Fig. 22.30(b) is courtesy of National Semiconductor Corporation.)



Hip implant-cyclic loading from walking.

Adapted from Fig. 22.26(b), *Callister 7e.* 

### Ductile vs. brittle failure

fracture is usually

desirable!

Classification: Very Moderately Fracture Brittle **Ductile** Ductile behavior: Adapted from Fig. 8.1, Callister 7e. %AR or %EL Large Moderate Small Ductile Ductile:

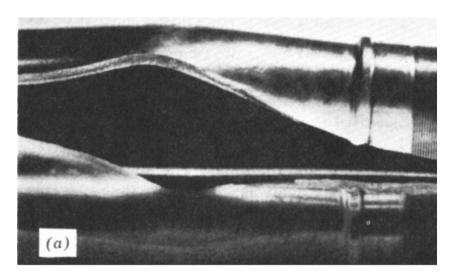
warning before fracture

No warning

## Example: failure of a pipe

#### Ductile failure:

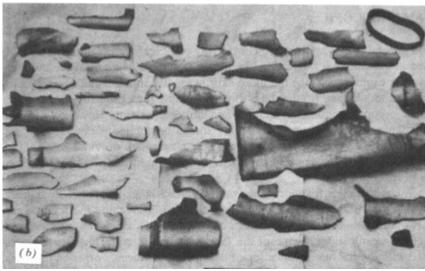
- --one piece
- --large deformation



#### • Brittle failure:

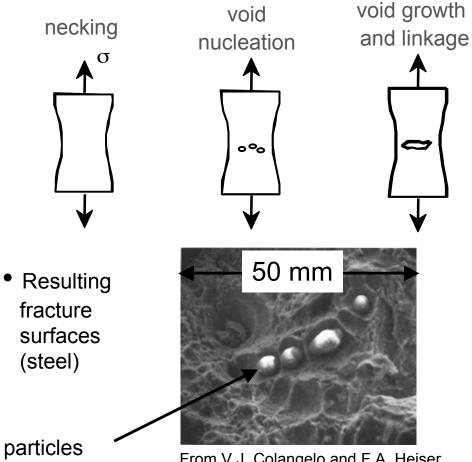
- --many pieces
- --small deformation

Figures from V.J. Colangelo and F.A. Heiser, *Analysis of Metallurgical Failures* (2nd ed.), Fig. 4.1(a) and (b), p. 66 John Wiley and Sons, Inc., 1987. Used with permission.



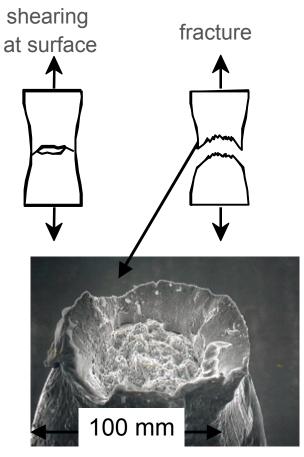
## Moderately ductile failure

#### Evolution to failure:



particles serve as void nucleation sites.

From V.J. Colangelo and F.A. Heiser, *Analysis of Metallurgical Failures* (2nd ed.), Fig. 11.28, p. 294, John Wiley and Sons, Inc., 1987. (Orig. source: P. Thornton, *J. Mater. Sci.*, Vol. 6, 1971, pp. 347-56.)



Fracture surface of tire cord wire loaded in tension. Courtesy of F. Roehrig, CC Technologies, Dublin, OH. Used with permission.

## Ductile vs. brittle failure



cup-and-cone fracture

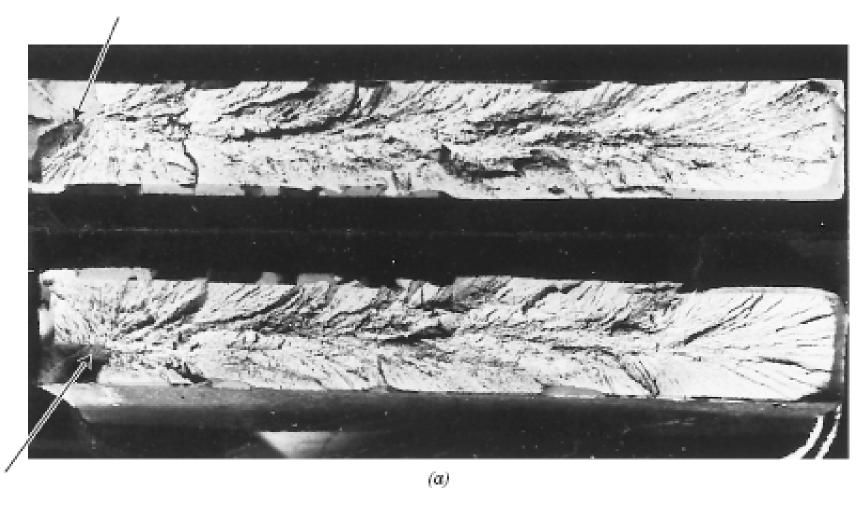


brittle fracture

Adapted from Fig. 8.3, Callister 7e.

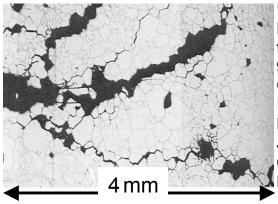
### Brittle failure

### Arrows indicate point at which failure originated



### Brittle fracture surfaces

Intergranular (between grains)



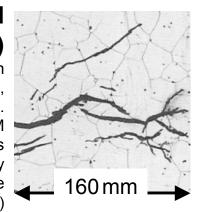
304 S. Steel (metal)

Reprinted w/permission from "Metals Handbook", 9th ed, Fig. 633, p. 650. Copyright 1985, ASM International, Materials Park, OH. (Micrograph by J.R. Keiser and A.R. Olsen, Oak Ridge National Lab.)

Intragranular (within grains)

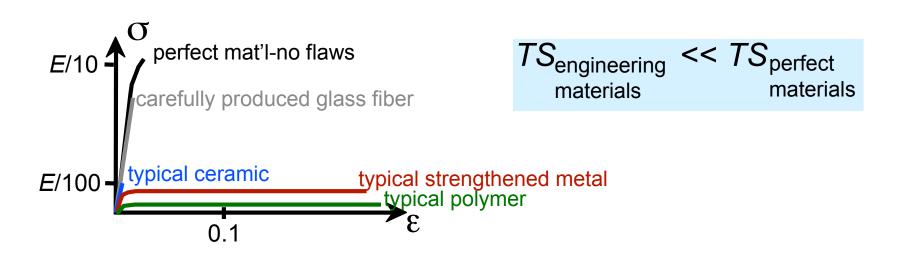
316 S. Steel (metal)

Reprinted w/ permission from "Metals Handbook", 9th ed, Fig. 650, p. 357. Copyright 1985, ASM International, Materials Park, OH. (Micrograph by D.R. Diercks, Argonne National Lab.)



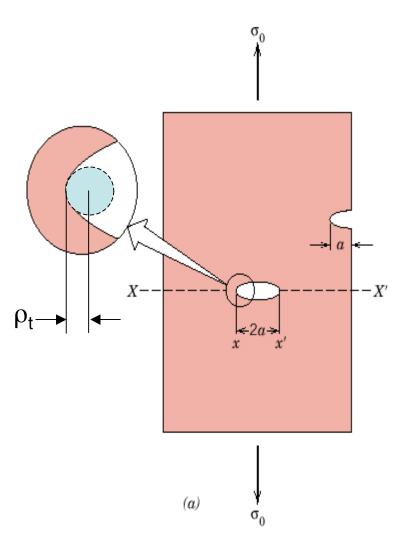
### Ideal vs. real materials

• Stress-strain behavior (Room *T*):



Flaws are Stress Concentrators!

### Ideal vs. real materials



Griffith Crack

$$\sigma_m = 2\sigma_0 \left(\frac{a}{\rho_t}\right)^{1/2} = K_t \sigma_0$$

where

 $\rho_t$  = radius of curvature  $\sigma_o$  = applied stress  $\sigma_m$  = stress at crack tip

### Crack propagation

#### Crack propagates if above critical stress

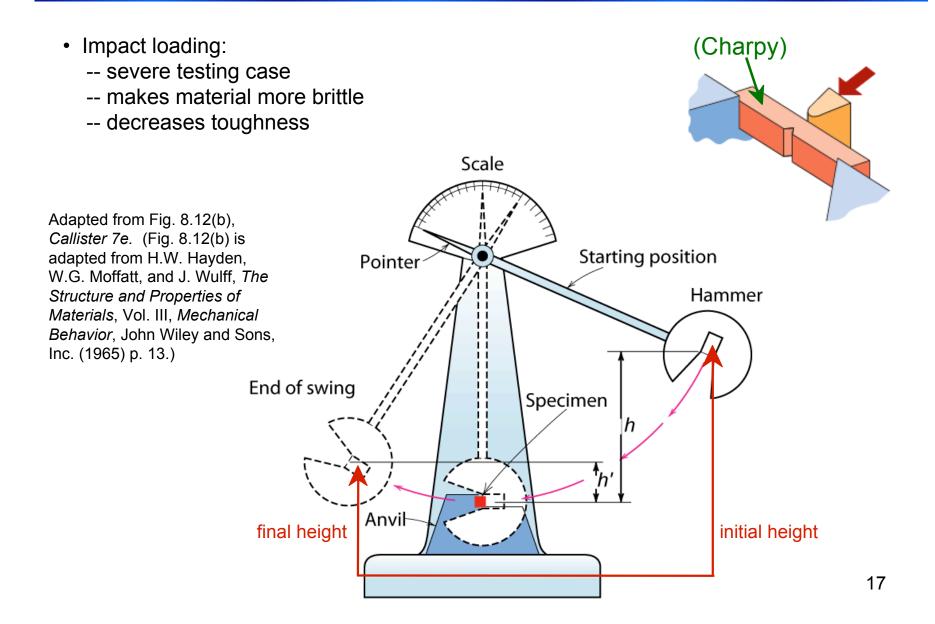
i.e., 
$$\sigma_m > \sigma_c$$
 
$$\sigma_c = \left(\frac{2E\gamma_s}{\pi a}\right)^{1/2}$$
 or  $K_t > K_c$ 

#### where

E = modulus of elasticity  $\gamma_s$  = specific surface energy a = one half length of internal crack  $K_c$  =  $\sigma_c/\sigma_0$ 

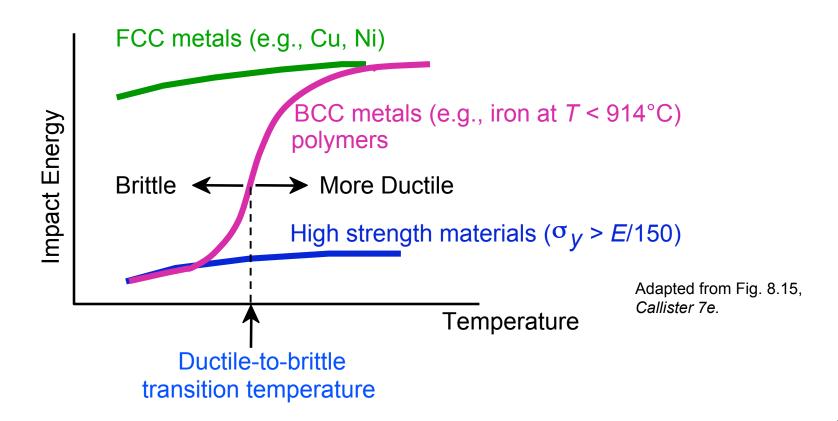
For ductile => replace  $\gamma_s$  by  $\gamma_s + \gamma_p$ where  $\gamma_p$  is plastic deformation energy

### Impact testing



### Temperature effects

- Increasing temperature...
  - --increases %EL and K<sub>c</sub>
- Ductile-to-Brittle Transition Temperature (DBTT)...



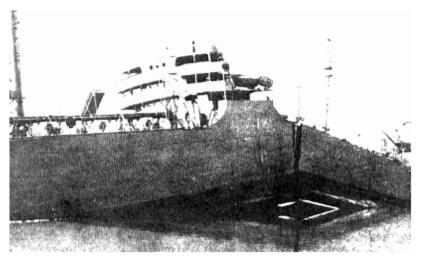
## Stay above DBTT

#### Pre-WWII: The Titanic



Reprinted w/ permission from R.W. Hertzberg, "Deformation and Fracture Mechanics of Engineering Materials", (4th ed.) Fig. 7.1(a), p. 262, John Wiley and Sons, Inc., 1996. (Orig. source: Dr. Robert D. Ballard, *The Discovery of the Titanic.*)

### • WWII: Liberty ships



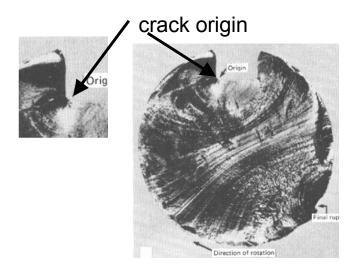
Reprinted w/ permission from R.W. Hertzberg, "Deformation and Fracture Mechanics of Engineering Materials", (4th ed.) Fig. 7.1(b), p. 262, John Wiley and Sons, Inc., 1996. (Orig. source: Earl R. Parker, "Behavior of Engineering Structures", Nat. Acad. Sci., Nat. Res. Council, John Wiley and Sons, Inc., NY, 1957.)

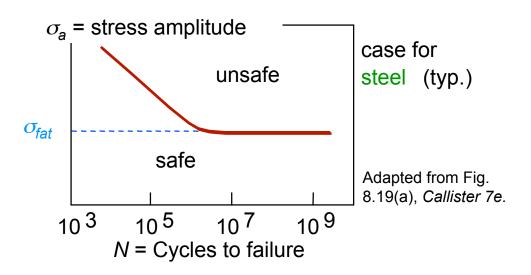
Problem: Used a type of steel with a DBTT ~ Room temp.

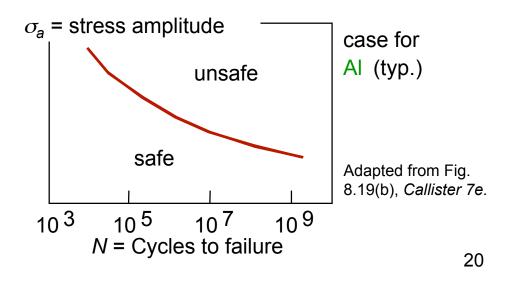
### Fatigue

• Fatigue limit,  $\sigma_{fat}$ :
--no fatigue if  $\sigma < \sigma_{fat}$ 

 Sometimes, the fatigue limit is zero!

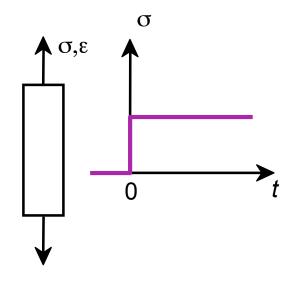






### Creep

#### Sample deformation at a constant stress ( $\sigma$ ) vs. time



Primary  $\Delta t$  Tertiary

Secondary

Instantaneous deformation

Time, t

Primary Creep: slope (creep rate) decreases with time.

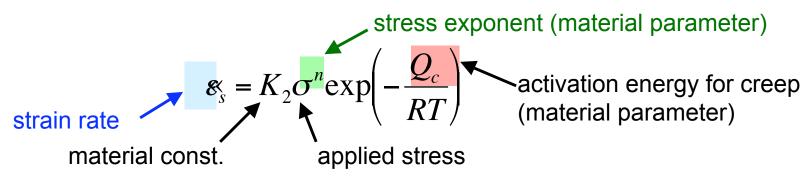
Secondary Creep: steady-state i.e., constant slope.

Tertiary Creep: slope (creep rate) increases with time, i.e. acceleration of rate.

Adapted from Fig. 8.28, *Callister 7e.* 

### Secondary creep

Strain rate is constant at a given T, σ
 -- strain hardening is balanced by recovery



Strain rate increases for higher *T*, σ

