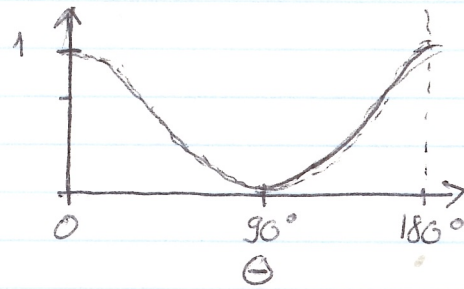
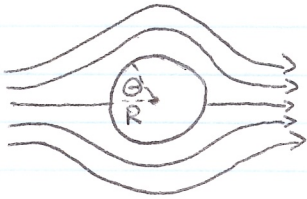


(1)

Potential Flow around Cylinder (crossflow)

Pressure coeff.
(dimensionless pressure)

$$C_p \equiv \frac{\Delta P}{\frac{1}{2} \rho U^2} = [1 - 4 \sin^2 \theta] \quad \text{based on potential flow theory}$$



$$\Delta P = p - p_{\infty}$$

p ... mean surface pressure

p_{∞} stream static pressure

Deen Eq. 9.3-16

$$\Delta P \equiv P(R, \theta)$$

$$= \frac{F_{\text{pressure drag}}}{RL}$$

(1) The pressure drag is given by:

$$C_{D,p} = \int_0^{\pi} C_p \cos \theta \, d\theta$$

For the idealized potential flow situation:

$$C_{D,p} = \int_0^{\pi} [1 - 4 \sin^2 \theta] \cos \theta \, d\theta = 0$$

This is an unrealistic result for the pressure drag coefficient. It is a consequence of the balanced pressures between the upstream and the downstream on the cylinder surface.

Data fit for high Re and smooth surfaces with laminar BL (laminar regime)

pressure coeff:

$$C_p = \begin{cases} 1 - \frac{8}{3} \sin^2 \theta & \text{for } 0 \leq \theta \leq \frac{\pi}{3} \\ -1 & \text{for } \frac{\pi}{3} \leq \theta \leq \pi \end{cases} \quad \begin{array}{l} \text{HW 6} \\ \text{Problem} \\ 6-6 \end{array}$$

=> pressure drag $\pi/3$

$$C_{D,P} = \int_0^{\pi/3} \left(1 - \frac{8}{3} \sin^2 \theta\right) \cos \theta d\theta - \int_{\pi/3}^{\pi} \cos \theta d\theta$$

$$= \frac{2}{\sqrt{3}}$$

used $\int_a^b \sin^2 \theta \cos \theta d\theta$
 $= \frac{\sin^3 \theta}{3} \Big|_a^b$

=> The pressure drag in stream direction is

$$F_p = (2RL) \left(\frac{\rho U^2}{2}\right) C_{D,P} ; \quad 2RL \dots \text{projected area of cylinder affected by cross flow}$$

$$= \frac{2}{\sqrt{3}} RL \rho U^2$$

The plot above demonstrates that there is a critical angle $\theta \neq \frac{\pi}{2}$ at which the functional behavior of the pressure drag is significantly changed. The reason lays in the separation of the boundary layer.

For a cylinder in crossflow, the boundary layer separates for

$$Re \leq 1$$

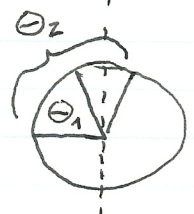
at ≈ 60 degrees

$$Re > 5 \times 10^5$$

at 107.7 degrees

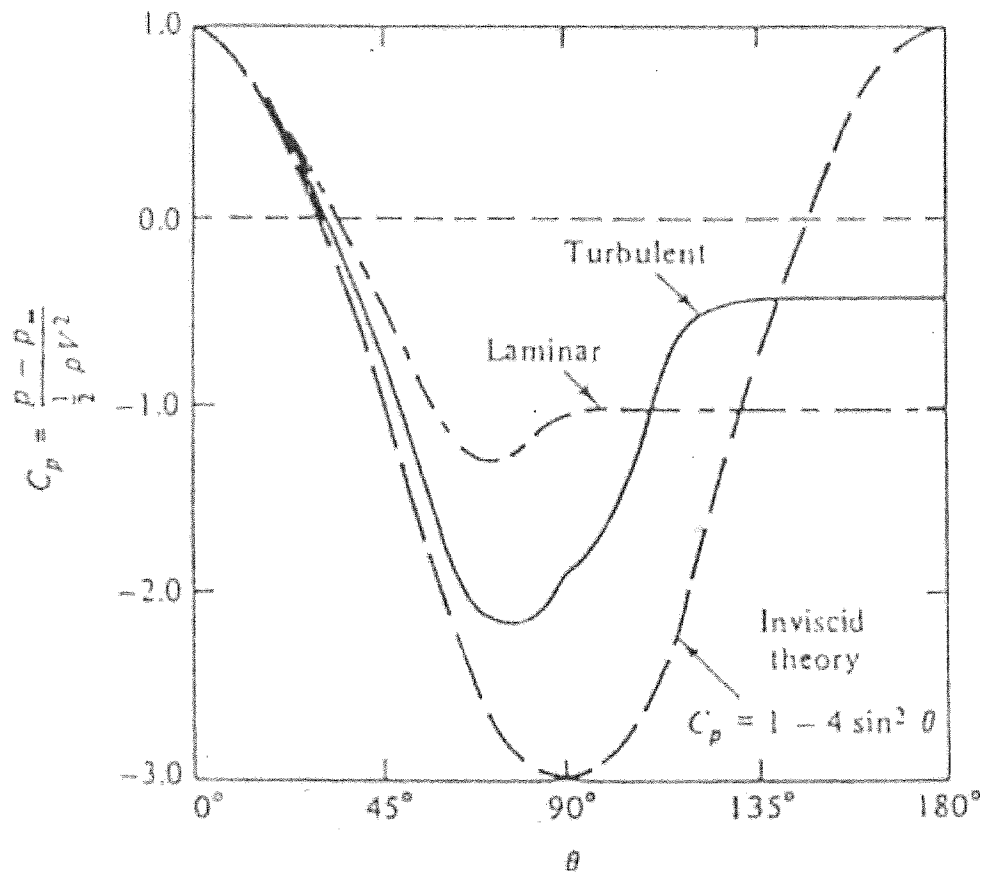
$\theta_1 = 60^\circ \Rightarrow$ laminar boundary layer

$\theta_2 = 107.7^\circ \Rightarrow$ turbulent boundary layer

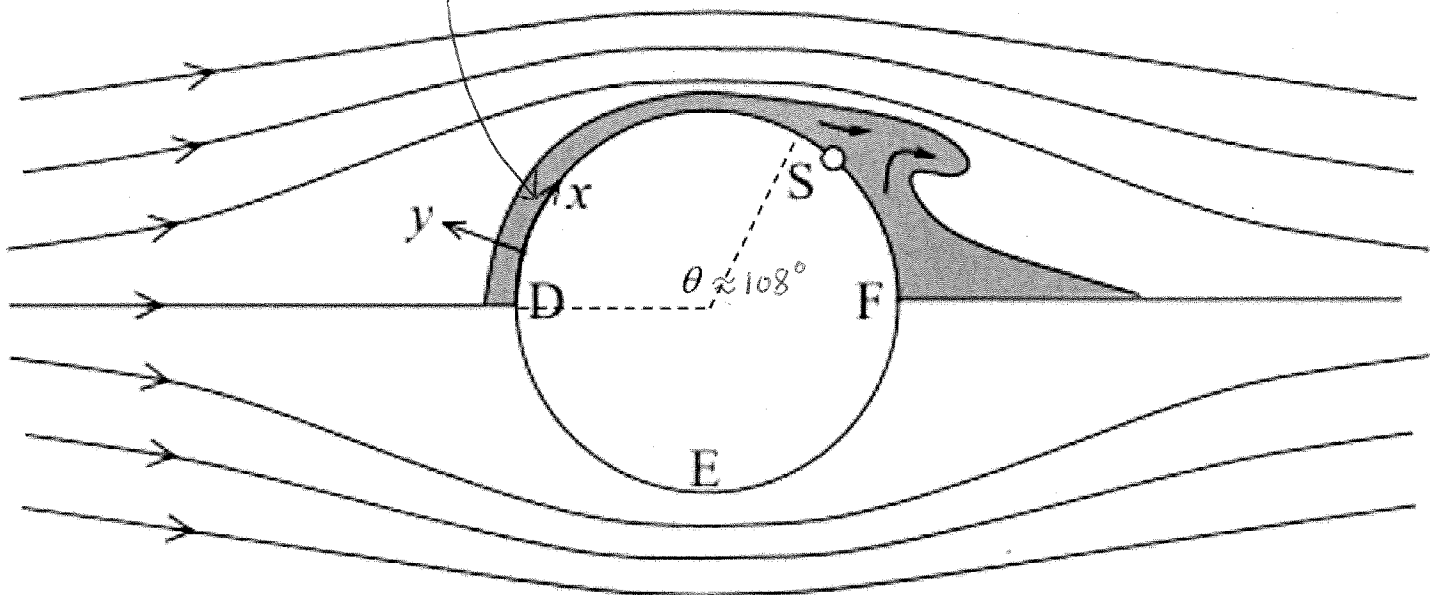


Crossflow around Cylinder

(29)



as $\theta \approx 108^\circ$, the boundary layer is turbulent



Boundary Layer Separation at θ

Laminar BL: $\theta \approx 80$ deg. (smooth cylinder surface)

Turbulent BL: $\theta = 107.7$ deg. (smooth cylinder surface)

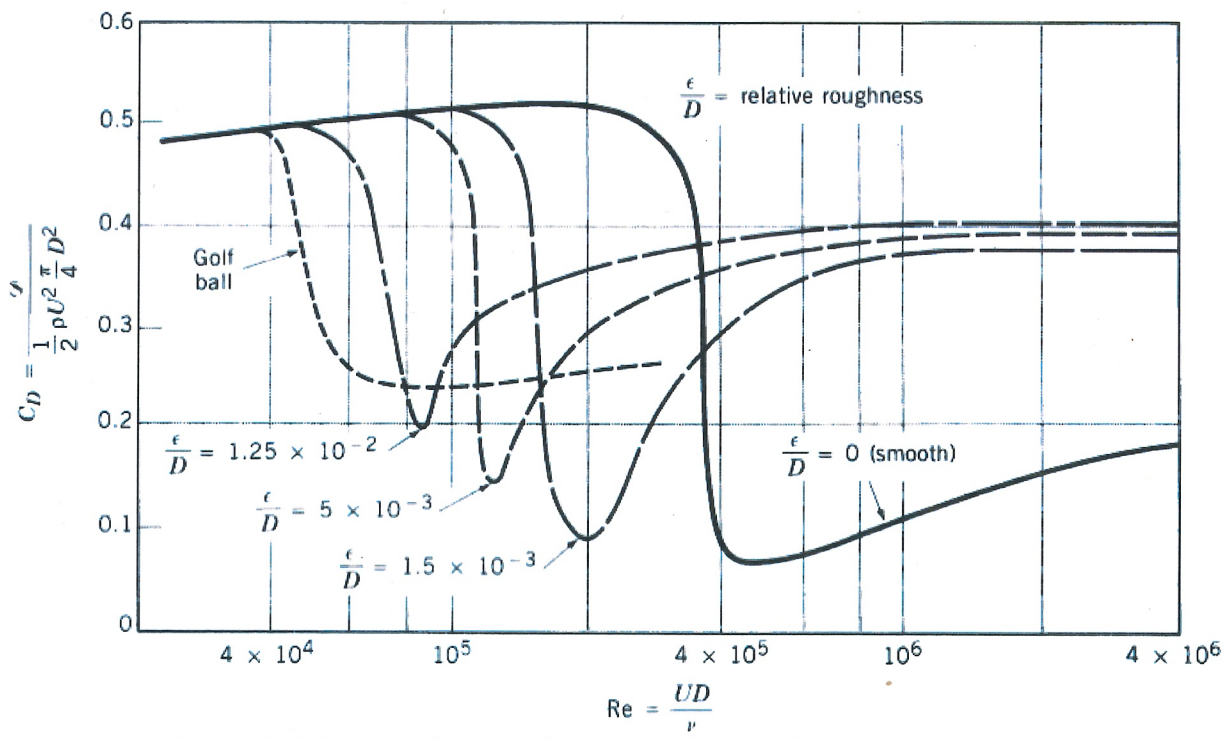
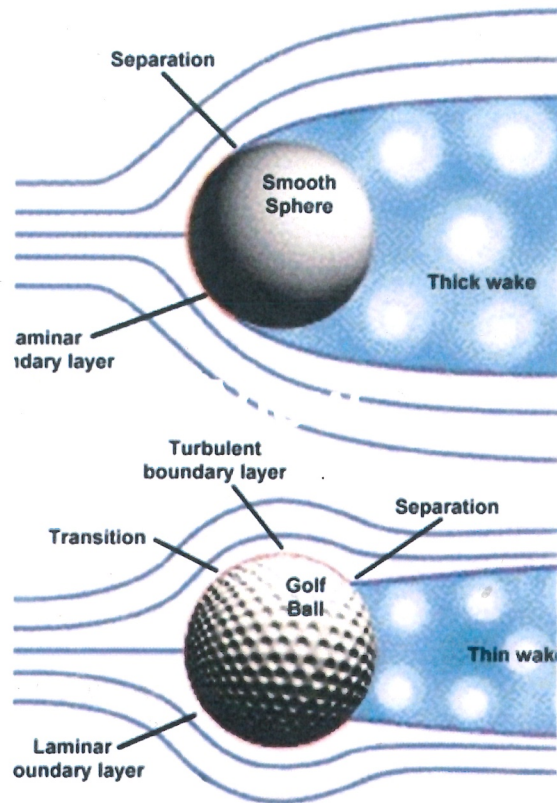


Figure 2. Effect of surface roughness on the drag coefficient of a sphere

https://www.mne.psu.edu/cimbala/me325web_Spring_2012/Labs/Drag/intro.pdf